

YUKON PROPELLER BY WIPAIRE AND HARTZELL

YUKON 208 PROPELLER INSTALLATION GUIDE

HARTZELL HC-E4N-KTV ()/GC11114



Revision	Pages Affected	Description	Date
1	All	Original Issue	3/1/2024
2	All	Complete format change, added all tech manual references, publications references, tooling references, etc	7/26/2024
3	2 18 28 41 42 43	Updated TOC to reflect pages added as noted below Added new photos for prop removal, removal wrenches Added new slide for oil seal reference Added REF for spinner screw misalignment Changed photo for spinner hole overlap Added new slide for cowl to spinner fitment issues	08/14/24
4	2 20 27 54-62 39-40 44	Updated TOC to reflect pages added as noted below Added page for deice brush block bracket removal Added note to install washer under beta ring puller foot Deleted all duplicated information now found on Wipaire drawing 1012329 REV B or later Added spinner IPC data from REV 26 of Hartzell Manual 127 Removed various hardware and part identifications now found in various Hartzell manuals.	9/4/24



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REQUIRED TOOLING



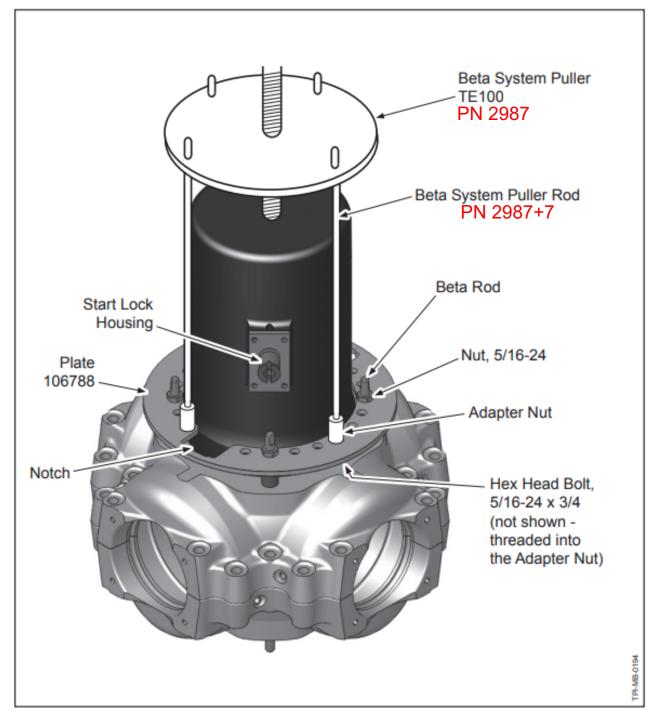
BETA RING PULLER PN CST-2987

CST-2987

PULLER, BETA RING



BETA RING PULLER PN CST-2987



Using the Plate Kit 106804 Figure 1

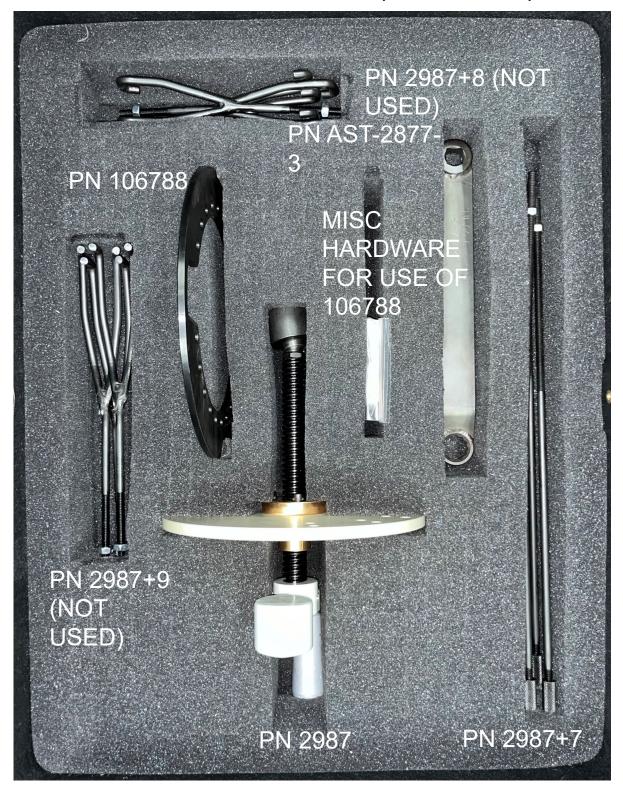


TORQUE WRENCH ADAPTER PN AST-2877-3



AST-2877-3
ADAPTER, TORQUE WRENCH

WPC00419 YUKON PROP TOOL KIT (RENTAL UNIT)





REQUIRED TECHNICAL DATA



APPLICABLE TECHNICAL DATA **AVAILABLE AT WWW.HARTZELLPROP.COM

-PROPELLER OWNERS MANUAL 147 -METAL SPINNER MAINTENANCE MANUAL 127

Manual No. 147 61-00-47 Revision 20 July 2023



Propeller Owner's Manual and Logbook

Models: HC-E(4,5)()-3() HC-E(4,5)()-5() HC-E5A-2()

Lightweight Turbine Propellers with Composite Blades

Hartzell Propeller Inc.

Hartzell Propeller Inc.
One Propeller Place
Piqua, OH 45356-2634 U.S.A.
Ph: 937-778-4200 (Hartzell Propeller Inc.)
Ph: 937-778-4379 (Product Support)
Product Support Fax: 937-778-4215

Manual No. 127 61-16-27 August 2024



Metal Spinner Maintenance Manual

Hartzell Propeller LLC One Propeller Place Piqua, Ohio 45356-2634 U.S.A. Phone: 937.778.4200 Fax: 937.778.4215

CURRENT REVISIONS AVAILABLE AT



APPLICABLE TECHNICAL DATA **AVAILABLE AT WWW.HARTZELLPROP.COM

-COMPOSITE BLADE FIELD MAINTENANCE MANUAL 170 -ICE PROTECTION SYSTEM MANUAL 180

Manual No. 170 61-13-70 Revision 21 September 2023



Bantam Composite

Composite Blade Field Maintenance and Minor Repair Manual

N-shank Composite

NC10320()

Legacy Composite B7421(K) ()7690() 7890K F8190K E9193(B.K) M10083(K) A10460(E)(K) LM10585(A)(N)(B,K)+4 M10877K E10950P(C)(B,K) F11990K E12902K E12903() E13890K 108MH92 138MH91

N7605(B,K)() ()75A01() N7605C() L76A01()X() N76M05C()X ()79A06X() N7893() N()8301() Raptor Composite N()8302() 75C()08() N()8304() 76C()03() NM8410() 76C()04() NM8411() 78D01() NC8834() ()79C03() NC9208() 80C()01() 84DB26() NC9405() NC10244() 86DB01() NC10245() 91D1(5,7)()

NC10445()
(J)NC10904()

G-shank Composite
GC1111(4,5)()

100DD44()

Hartzell Propeller Inc. One Propeller Place Piqua, Ohio 45356-2634 U.S.A. Phone: 937.778.4200 Fax: 937.778.4215 Manual No. 180 30-61-80 Revision 41 September 2023



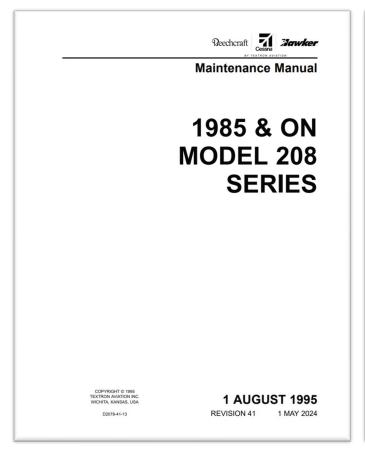
Ice Protection System Manual

Hartzell Propeller Inc. One Propeller Place Piqua, Ohio 45356-2634 U.S.A. Phone: 937.778.4200 Fax: 937.778.4215

CURRENT REVISIONS AVAILABLE AT WWW.HARTZELLPROP.COM



-CESSNA 208 MAINTENANCE MANUAL -CESSNA 208 ILLUSTRATED PARTS CATALOG

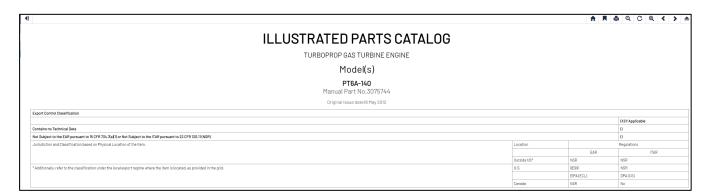




AVAILABLE BY SUBSCRIPTION ONLY FROM TXTAV



-PRATT AND WHITNEY PT6A-140 IPC AND SERVICE MANUALS



Issued 04 June 2012

MAINTENANCE MANUAL TURBOPROP ENGINE MODEL(S) PT6A-140 ENGINES Manual Part No.3075742

AVAILABLE BY SUBSCRIPTION ONLY



-PRATT AND WHITNEY PT6A-114A IPC AND SERVICE MANUALS

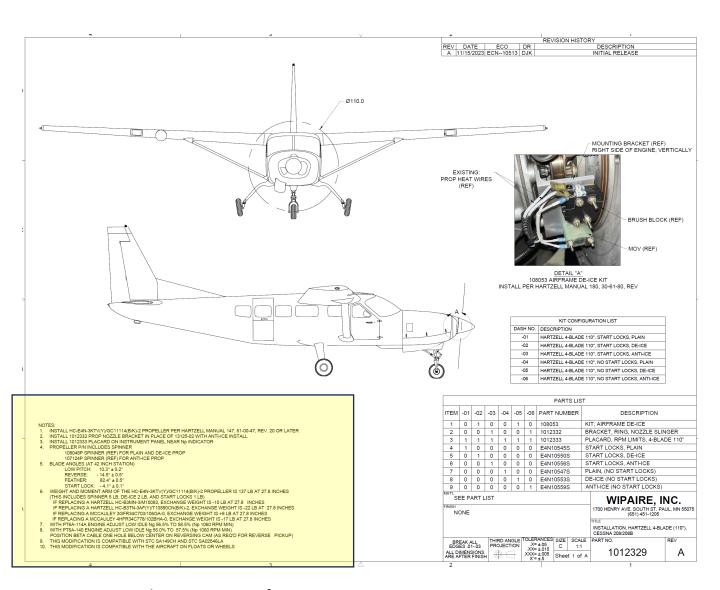
ILLUSTRATED PARTS CATALOG TURBOPROP GAS TURBINE ENGINE			
Model(s)			
PT6A-114/-1/4A/-135/-135A Manual Part No.3043514			
Original issue date31 January 1987			
Export Control Classification			
			(X) if Applicable
Contains no Technical Data			0
Not Subject to the EAR pursuant to 15 CPR 734.7(a)(1) or Not Subject to the ITAR pursuant to 22 CPR 120.11(NSR)			0
Jurisdiction and Classification based on Physical Location of the Item.			Regulations
		EAR	ITAR
Outside US*	JS* N	NSR .	NSR
* Additionally, refer to the classification under the local export regime where the item is located, as provided in the grid. U.S.	91	E991	NSR
			DPA(CG)
Canada	N	4SR	No

MAINITENANCE MANULAL				
MAINTENANCE MANUAL				
TURBOPROP GAS TURBINE ENGINE				
Model(s)				
PT6A-114/-114A/-135/-135A Manual Part No.3043512				
Original Issue date31 October 1997				
The contents of this manual have been examined and found acceptable to the Minister in meeting the requirements of an Engine Maintenance Manual for the Pratt & Whitney Canada PT8A-114/-1135/-135A engine, as required by the Canadian Airworthiness Manual, Chapter 533.4 "Instructions for Continued Airworthiness". [2024/07/88/72]				
Export Control Classification				
O/M Applicable				
	Contains no Technical Data (1)			
Not Subject to the EAR pursuant to 15 CFR 73-47,a[1] or Not Subject to the ITAR pursuant to 22 CFR 120.11 (NSR) ()				
Jurisdiction and Classification based on Physical Location of the Item.	Location	FAR	Regulations ITAR	
	Outside US*	NSR	NSR	
	U.S.	9E991	NSR	
* Additionally, refer to the classification under the local export regime where the item is located, as provided in the grid,	0,5,	EIPA(ECL)	DPA(CG)	
	Canada	NSR	No.	
	Canada	Inon	NO	

AVAILABLE BY SUBSCRIPTION ONLY



WIPAIRE INSTALLATION DRAWING 1012329



Pay particular attention of notes shown on LH side of drawing



HC-E4N-3KTV(Y)/GC11114(B/K)-2 PROPELLER

KIT CONFIGURATION LIST		
DASH NO.	DESCRIPTION	
-01	HARTZELL 4-BLADE 110", START LOCKS, PLAIN	
-02	HARTZELL 4-BLADE 110", START LOCKS, DE-ICE	
-03	HARTZELL 4-BLADE 110", START LOCKS, ANTI-ICE	
-04	HARTZELL 4-BLADE 110", NO START LOCKS, PLAIN	
-05	HARTZELL 4-BLADE 110", NO START LOCKS, DE-ICE	
-06	HARTZELL 4-BLADE 110", NO START LOCKS, ANTI-ICE	

WIPAIRE KIT BILLS OF MATERIALS

· 1012329-01

- 1012333 Placard, Caravan, RPM Limits, Hartzell 4-Blade
- E4N10545S Prop, Hartzell, Yukon, Four Blade Carbon Composite For Cessna Caravan, No Ice Protection, Includes Pitch Locks

1012329-02

- 108053 Kit, Airframe De-ice
- 1012333 Placard, Caravan, RPM Limits, Hartzell 4-Blade
- E4N10550S Yukon, Four Blade Carbon Composite For Cessna Caravan, Electrically Heated De-Ice Protection, Includes Pitch Locks

1012329-03

- 1012332 Bracket, TKS, Nozzle
- 10123333 Placard, Caravan, RPM Limits, Hartzell 4-Blade
- E4N10556S Prop, Hartzell, Yukon, Four Blade Carbon Composite For Cessna Caravan, TKS Anti-Ice Protection, Includes Pitch Locks

1012329-04

- 1012333 Placard, Caravan, RPM Limits, Hartzell 4-Blade
- E4N10547S Prop, Hartzell, Yukon, Four Blade Carbon Composite For Cessna Caravan, No Ice Protection, No Pitch Locks

1012329-05

- 108053 Kit, Airframe De-ice
- 1012333 Placard, Caravan, RPM Limits, Hartzell 4-Blade
- E4N10553S Prop, Hartzell, Yukon, Four Blade Carbon Composite For Cessna Caravan, Electrically Heated De-Ice Protection, No Pitch Locks

1012329-06

- 1012332 Bracket, TKS, Nozzle
- 1012333 Placard, Caravan, RPM Limits, Hartzell 4-Blade
- E4N10559S Prop, Hartzell, Yukon, Four Blade Carbon Composite For Cessna Caravan, TKS Anti-Ice Protection, No Pitch Locks



ORIGINAL PROPELLER REMOVAL



REFERENCE **208 MAINTENANCE MANUAL 61-11-00** FOR MCCAULEY PROPELLER REMOVAL.

REFERENCE **208 MAINTENANCE MANUAL 61-10-00** FOR HARTZELL PROPELLER REMOVAL

Safely secure the prop to be removed with a forklift or hoist in position as shown.

Remove existing propeller using 208 Maintenance Manual, Chapter 61 as applicable

Clean up all spilled oil, search for FOD and overall condition of the area after old prop is removed.

IMPORTANT-Record removed propeller total times (TSN, TSO, etc..) and enter into appropriate logbook. Retain this logbook with removed propeller.

Note: Reference below for wrench adapters for each application







AST 2877 Hartzell



B-5588 McCauley





TKS EQUIPPED AIRCRAFT MUST LOOSEN OR REMOVE TKS
NOZZLE TO PREVENT DAMAGE DURING PROPELLER REMOVAL









• DEICE (ELECTRIC BOOT)
EQUIPPED PROPS MAY
BENEFIT FROM DEICE
BRUSH BLOCK BRACKET
REMOVAL PRIOR TO
REMOVAL OF THE
PROPELLER FOR EASE OF
ACCESSING PROP
FLANGE NUTS.



YUKON PROPELLER INSTALLATION



PROPELLER INSTALLATION SHALL BE PERFORMED USING HARTZELL, CESSNA, AND APPLICABLE PRATT AND WHITNEY DATA. WIPAIRE HAS CHOSEN TO PROVIDE TIPS ON KEY POINTS TO WATCH OUT FOR DURING PROPELLER MOUNTING TO ENGINE. CITING MANUFACTURERS DATA ALONG THE WAY.



APPLICABLE TECHNICAL DATA **AVAILABLE AT WWW.HARTZELLPROP.COM

-PROPELLER OWNERS MANUAL 147 SECTION 3 INSTALLATION AND REMOVAL PROCEDURES

Manual No. 147 61-00-47 Revision 20 July 2023



Propeller Owner's Manual and Logbook

Models: HC-E(4,5)()-3() HC-E(4,5)()-5() HC-E5A-2()

Lightweight Turbine Propellers with Composite Blades

Hartzell Propeller Inc. One Propeller Place Piqua, OH 45356-2634 U.S.A.

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HARTZELL PROPELLER OWNER'S MANUAL INSTALLATION AND REMOVAL - CONTENTS A. General 3-5 B. Tooling..... C. Consumables... D. Expendables ... 2. Pre-Installation..... A. Inspection of Shipping Package..... B. Uncrating..... D. Reassembly of a Propeller Disassembled for Shipment. 3-11 3. Propeller Mounting Hardware and Torque Information....... 3-11 A. Propeller Mounting Hardware3-11 B. Torque Information 4. Propeller Assembly Installation..... A. Precautions... B. Installing HC-E(4,5)()-3() except HC-E5P-3 and C. Installing the HC-E5P-3 Propeller on the Aircraft Engine 3-23 D. Installing the HC-E5W-3() Propeller on the Aircraft Engine..... E. Installing the HC-E(4,5)N-5KL Propeller on the Aircraft Engine..... F. Installing HC-E5A-2 Propeller on the Aircraft Engine..... 3-39 G. Installing HC-E5(A,B)-5A Propeller on the Aircraft Engine.. 3-45 B. For Propeller Model HC-E5A-2, Installing the Striker Plates (Pilatus part)3-51 C. For Propeller Model HC-E5A-2, Measuring the 3-53

D. Installing the Spinner Dome......

INSTALLATION AND REMOVAL 61-00-47 Re

CURRENT REVISIONS AVAILABLE AT WWW.HARTZELLPROP.COM



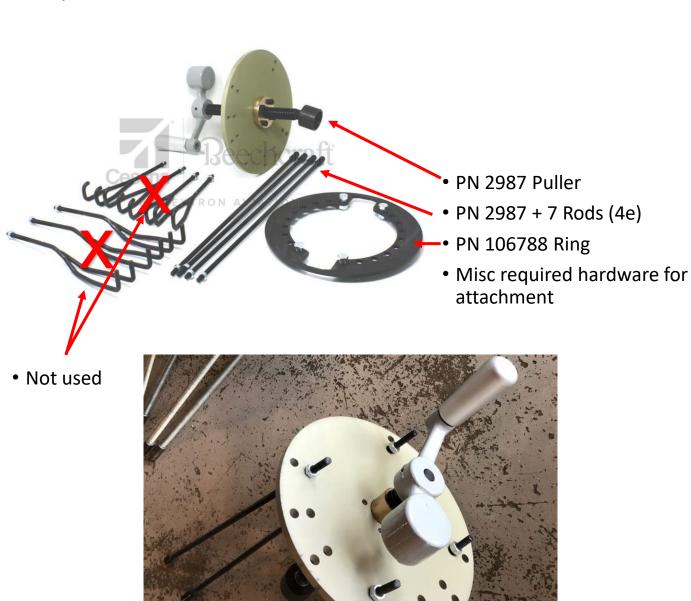
... 3-53

The YUKON propeller will consist of a KIT containing the propeller that was ordered to your configuration (plain, deice, anti ice, start locks, etc..) and a spinner dome and spinner bulkhead, spacers, and attach hardware as a single part number, boxes 1 of 2 and 2 of 2. Optional small items may be associated with your installation such as airframe deice kit, TKS bracket, etc... Verify all paperwork serial numbers match the actual propeller that was shipped.



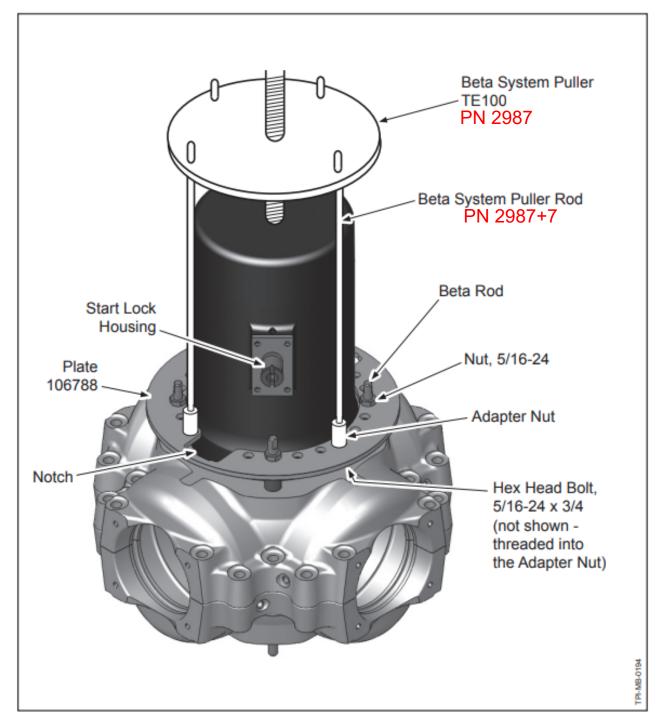


Using the CST-2987 Beta Ring decompression tool, configure the puller as shown below.





BETA RING PULLER PN CST-2987



Using the Plate Kit 106804 Figure 1



Install Beta Ring decompressing tool to propeller utilizing plate PN 106788 and related misc hardware as shown.







 Install NAS1149F0963P (AN960-916) washer between beta ring and prop as shown to provide stability to swivel.











Install new prop flange O ring seal, this will be in your new prop kit.

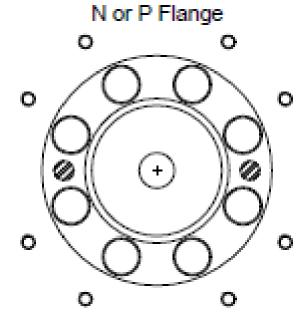






Important! Locate the orientation of the propeller pins on the prop shaft.

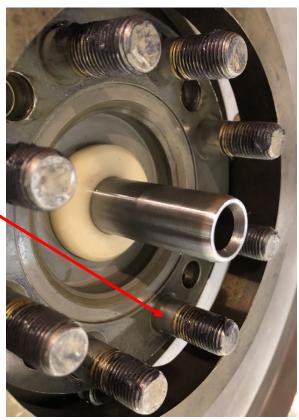
The PT6 on this installation utilizes the *N flange* as referenced in Hartzell Owners Manual 147, Figure 3-5





Apply Hartzell provided anti-seize compound PN A-3338 to threaded studs on prop. This is typically provided with your new propeller.

Align the Yukon prop in the correct orientation to the prop shaft dowell pins

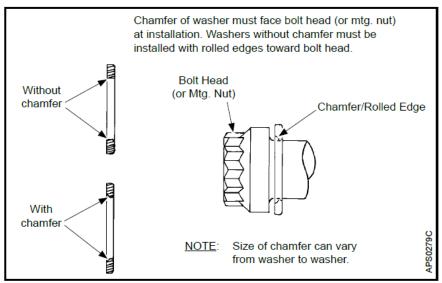






Make note of washer, nut, and O ring reference as noted in Hartzell Owners Manual 147, Chapter 3, Figure and Table 3-1 as referenced below. Orientate washers accordingly

HARTZELL PROPELLER OWNER'S MANUAL 147



Mounting Bolt/Nut and Washer Figure 3-1

Flange	O-ring	Bolt/Stud	Washer	Nut	Misc
A - except E5A-2	C-3317-239-2	B-3347	A-2048-2	n/a	n/a
E5A-2	C-3317-239-2	B-7435	A-2048-2	C-6006	n/a
В	C-3317-239-2	B-3347	A-2048-2	n/a	n/a
N - except E(4,5)N-5KL and E4N-3KTV(Y)	C-3317-230	B-3339-1	A-2048-2	n/a	n/a
E(4,5)N-5KL and E4N-3KTV(Y)	C-3317-230	103560	A-2048-2	C-6006	n/a
P - except E5P-3	C-3317-230	B-3347	A-2048-2	n/a	n/a
E5P-3	C-3317-230	103560	A-2048-2	C-6006	n/a
E5W-3 - except E5W-3Y	C-3317-230	104720	B-7624	C-6006	106943 Spacer B-3868-S60 Screw
E5W-3Y	C-3317-230	104720	B-7624	C-7458	106943 Spacer B-3868-S60 Screw

Propeller/Engine Flange O-rings and Mounting Hardware
Table 3-1



Torque per the installation instructions found in Hartzell Owners Manual 147, Section 3, Table 3-2 utilizing torque adapter PN AST-2877-3. Torque per sequence found in Figure 3-3.

HARTZELL PROPELLER OWNER'S MANUAL

FOR A PROPELLER THAT DOES NOT USE A LUBRICATED (WET) TORQUE, THE MOUNTING

HARDWARE MUST BE CLEAN AND DRY TO PREVENT EXCESSIVE PRELOAD OF THE

MOUNTING FLANGE.

CAUTION 2: TORQUE VALUES WITH "WET" NOTED AFTER

THEM ARE BASED ON LUBRICATED THREADS WITH APPROVED ANTI-SEIZE COMPOUND

MIL-PRF-83483().

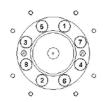
REFER TO FIGURE 3-2 FOR TORQUE READING CAUTION 3:

WHEN USING A TORQUE WRENCH ADAPTER.

A flange propeller mounting bolts (except E5A-2)	100-105 Ft-Lbs (136-142 N•m) Wet
E5A-2 flange propeller mounting nuts	120-130 Ft-Lbs (163-176 N•m) Wet
B flange propeller mounting bolts	100-105 Ft-Lbs (136-142 N•m) Wet
N flange propeller mounting bolts except HC-E(4,5)N-(3,5)K(L) and HC-E4N-3K(T)(V)(Y)	100-105 Ft-Lbs (136-142 N•m) Wet
N flange propeller mounting nuts HC-E(4,5)N-(3,5)K(L) and HC-E4N-3K(T)(V)(Y)	120-130 Ft-Lbs (163-176 N•m) Wet
P flange propeller mounting bolts except HC-E5P-3	100-105 Ft-Lbs (136-142 N•m) Wet
HC-E5P-3 propeller mounting nuts	120-130 Ft-Lbs (163-176 N•m) Wet
W flange propeller mounting nuts	120-125 Ft-Lbs (163-170 N•m)
Adapter Plate to Hub bolts -B-3384-4H	8-10 Ft-Lbs (10.8-13.5 N•m)
Slip Ring and adapter Plate Unit to Hub screws - A-2070-7	8-10 Ft-Lbs (10.8-13.5 N•m)
Bulkhead to Adapter Plate screws -B-3867-269	8-10 In-Lbs (0.9-1.1 N•m)
Pulley to Adapter Ring - Pilatus fastener	48 In-Lbs (5.4 N•m)
Balance weight screws or bolts- Aircraft quality #10-32 or AN-3()	30-36 In-Lbs (3.4-4.0 N•m)

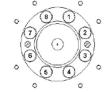
Torque Table Table 3-2

N, P or W Flange



SEQUENCE A

Use Sequence A for steps one and two.



SEQUENCE B

Use Sequence B for step three.

Step 1 - Torque all bolts/nuts to 40 Ft-Lbs (54 N•m). Step 3 - Torque all bolts/nuts Step 2 - Torque all bolts/nuts to 80 Ft-Lbs (108 N•m). to Table 3-2.

Torquing Sequence for Propeller Mounting Bolts/Nuts Figure 3-3









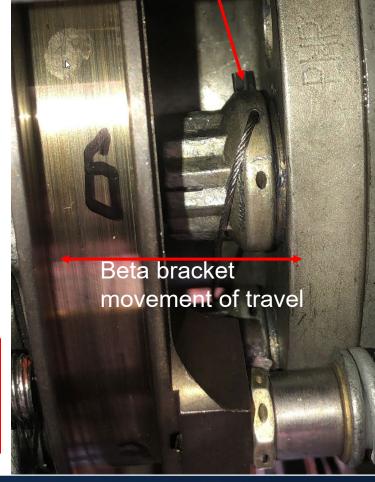


Safety wire or Safe-T-Cable to secure prop nuts after torque sequences. Do not allow pig tails or cable crimps to protrude outwards. This may result in undesirable contact with the beta bracket as it moves forward and aft. **Note: Highly recommended to use Safe-T-Cable over safety wire**



Incorrect

Correct



CAUTION:

THE BETA FEEDBACK COLLAR MUST NOT CONTACT ANY ENGINE COMPONENT OR MOUNTING BOLT SAFETY WIRE. THE BETA FEEDBACK MECHANISM COULD BE DAMAGED IF IT CONTACTED ANY STATIC ENGINE COMPONENT WHILE ROTATING.



SUGGESTED SAFETY CABLE TOOL WWW.DMCTOOLS.COM

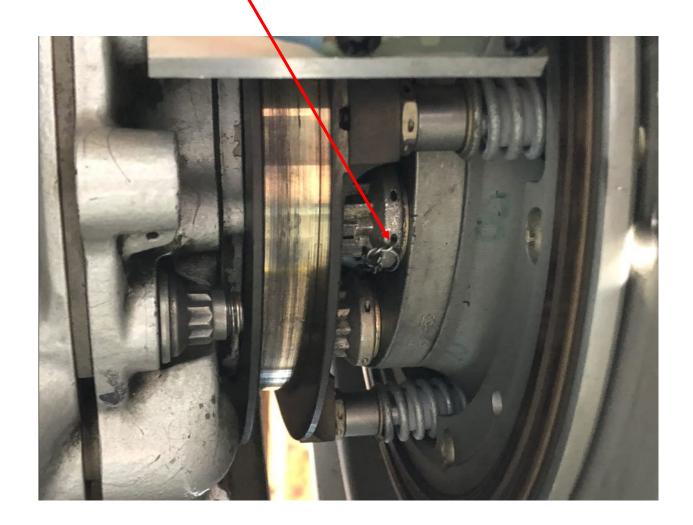


(8) Safety all propeller mounting nuts with 0.032 inch (0.81 mm) minimum diameter stainless steel wire or equivalent aircraft safety cable, two nuts for each safety.



Important-While you CAN use safety wire, it will be very difficult to access the nuts adequately and wire cleanly, let alone keep the wire away from beta ring.

Extremely tight area



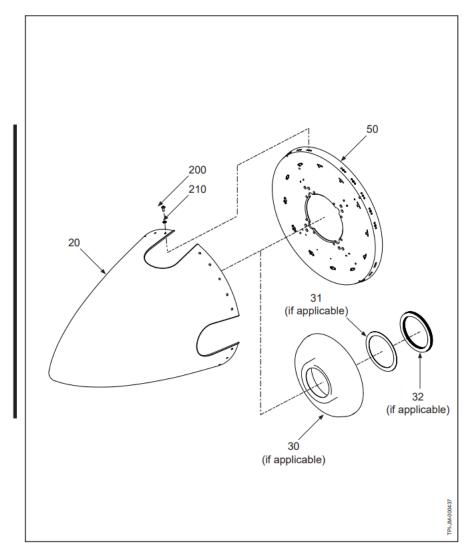


SPINNER DOME INSTALLATION



REFERENCE HARTZELL SPINNER PARTS BREAKDOWN PER HARTZELL MANUAL 127, REV 26 OR LATER

HARTZELL METAL SPINNER MAINTENANCE MANUAL



Example of a Four-Bladed Propeller Spinner -One-piece Dome without Filler Plates Figure 10.4-3

ILLUSTRATED PARTS LIST 61-16-27 Page 10.4-4 Rev. 23 Feb/24



REFERENCE HARTZELL SPINNER PARTS BREAKDOWN PER HARTZELL MANUAL 127, **REV 26 OR LATER**

TKS CONFIGURATION

- 1	10.4-3			- 1	l	
-	-10	107124(P)	SPINNER ASSEMBLY			
-	20	107196(P)	• DOME		1	
	30	C-5324	FORWARD BULKHEAD		1	
-	31	B-632	SHIM, SPINNER		15	
	50	107123(P)	BULKHEAD UNIT (INCLUDES BONDED BULKHEAD RING, SLINGER RING, AND FITTING)		1	
-	-59	B-3846-3	NUTPLATE, FIXED		16	
	-61	B-3849-3	• • NUTPLATE, FLOATING		16	
-	-62	B-3847-5	RIVET, 100° HEAD, .094 DIA, AL		64	
	-230	B-3384-4H	• • BOLT, 1/4-28, HEX HEAD		8	Y
-	-255	B-3837-0432	WASHER, CORROSION RESISTANT		8	Y
	200	B-3845-8	SCREW, 10-32, TRUSS HEAD (DOME-TO-BULKHEAD)		20	Y
ı	210	A-1020	WASHER, FIBER		20	Y
		1			ı	

DEICE AND PLAIN CONFIGURATION

		l	l l		
	10.4-3				
	-10	108048(P)	SPINNER ASSEMBLY		
	20	107196(P)	• DOME	1	
	30	C-5324	FORWARD BULKHEAD	1	
	31	B-632	SHIM, SPINNER	15	
	50	108049(P)	BULKHEAD UNIT	1	
	-59	B-3846-3	• • NUTPLATE, FIXED	16	
	-61	B-3849-3	• • NUTPLATE, FLOATING	16	
	-62	B-3847-5	RIVET, 100° HEAD, .094 DIA, AL	64	
	200	B-3845-8	SCREW, 10-32, TRUSS HEAD (DOME-TO-BULKHEAD)	20	Υ
	210	A-1020	WASHER, FIBER	20	Υ
_					
		l			



INSTALLATION OF SPINNER DOME IS PERFORMED PER HARTZELL OWNERS 143, SECTION 3, PAR 5 AS SHOWN BELOW

HARTZELL PROPELLER OWNER'S MANUAL 147

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		B. Installing HC-E(4,5)()-3() except HC-E5P-3 and	
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		C. Installing the HC-E5P-3 Propeller on the Aircraft Engine	
		D. Installing the HC-E5W-3() Propeller	
		on the Aircraft Engine	
		E. Installing the HC-E(4,5)N-5KL Propeller	
l		on the Aircraft Engine	
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_		G. Installing HC-E5(A,B)-5A Propeller on the Aircraft Engine 3-45	
	5.	Spinner Dome Installation	
		A. General	
		B. For Propeller Model HC-E5A-2, Installing the Striker Plates (Pilatus part)	
		C. For Propeller Model HC-E5A-2, Measuring the	
		Resistances	
		D. Installing the Spinner Dome	
		Dans 2.1	

INSTALLATION AND REMOVAL 61-00-47 Rev. 21 May/24

HARTZELL PROPELLER OWNER'S MANUAL 147

5. Spinner Dome Installation

CAUTION 1: TO PREVE

TO PREVENT DAMAGE TO THE BLADE AND BLADE PAINT, WRAP THE BLADE SHANKS IN SEVERAL LAYERS OF MASKING OR DUCT TAPE BEFORE INSTALLING THE SPINNER DOME. REMOVE THE TAPE AFTER THE SPINNER IS INSTALLED.

CAUTION 2: SPINNER

SPINNER DOME WILL WOBBLE IF NOT ALIGNED PROPERLY. THIS MAY AFFECT DYNAMIC BALANCE OF PROPELLER.

- A. General
 - The following instructions relate to Hartzell Propeller spinners only.
 - (a) In some cases, the airframe manufacturer produced the spinner assembly. Refer to the airframe manufacturer's manual for spinner dome installation instructions.
- B. For Propeller Model HC-E5A-2, Installing the Striker Plates (Pilatus part). Refer to Figure 3-15.
 - The striker plates must have a curvature to match that of the dome and must have a weight of not more than 0.458 ounces (13 grams).
 - (2) Using screws, part number B-3867-272, install the striker plates centered between the blades, using the mounting holes provided.

NOTE: Washers, part number B-3860-10L, are not used at the striker plate attachment points.

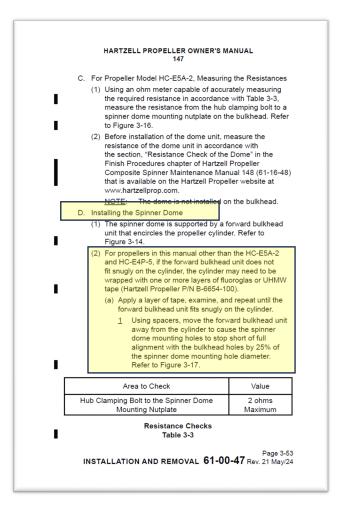
Page 3-51
INSTALLATION AND REMOVAL 61-00-47 Rev. 21 May/24

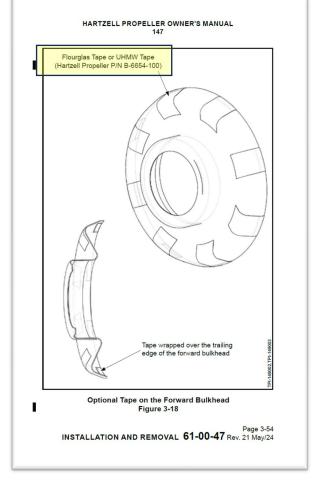


Installation of the FWD spinner bulkhead PN C-5324 onto hub as shown below may require Hartzell Propeller Tape PN B-6654-100 to be installed as shim material for shimming radially (side to side)

There is an optional use of this same tape between bulkhead and spinner dome as noted in Par D and Figure 3-18 shown below.

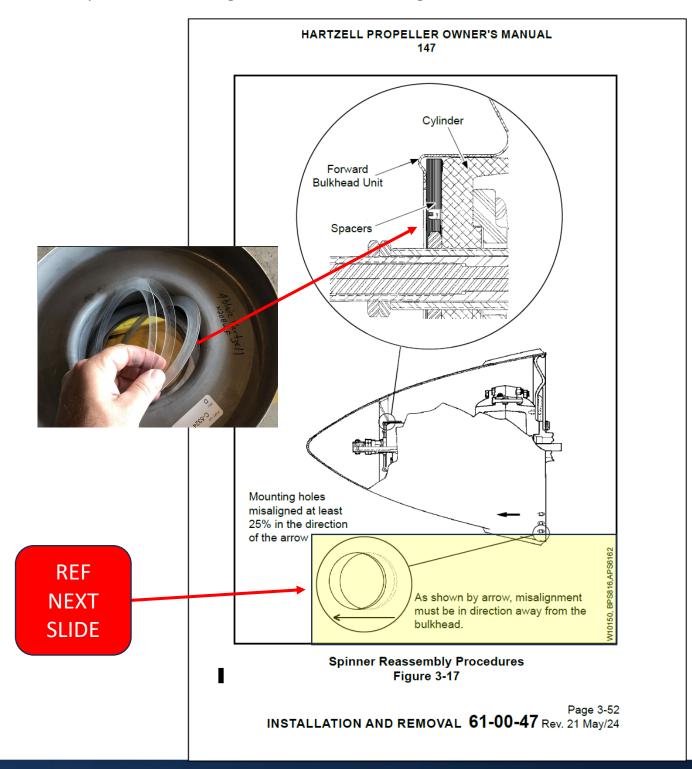






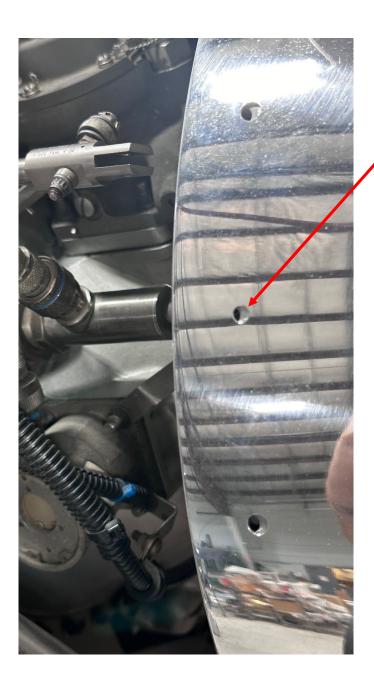


You are trying to achieve a fitment using supplied spacers with spinner to backplate hole misalignment as shown in Figure 3-17 Below.





Gently and evenly install the spinner. Once holes are aligned start with the center holes first in all four spinner quadrants.



Once these center four screws are installed, work outward on the remaining 4 screws.



NOTE: There may be slight spinner to nose cowl clearance issues. You may have to adjust the RH nose cowl bracket at this location to achieve sufficient space needed for clearance. You may also have to elongate holes slightly and reposition bracket if more clearance is needed. In the photo below, note the tightest area is typically just above oil cooler opening on RH cowl half







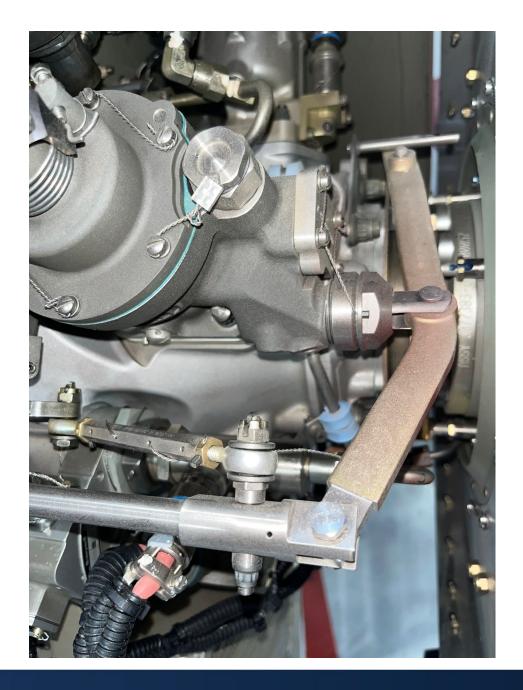


BETA FEEDBACK ARM



Installation of the reversing linkage arm is accomplished using **3 different data sources.**

- 1. Hartzell Owners Manual 147, Section 3 and Figure 3-7 for fitment of the block into the beta ring and ONTO the arm
- 2. Cessna Maintenance Manual 60-10-00 and 61-20-00
- 3. Pratt and Whitney Maintenance and Parts manual for applicable engine configuration (-114A or -140) Chapter 76 ENGINE CONTROLS





1. Hartzell Owners Manual 147, Section 3 and Figure 3-7 for fitment of the block into the beta ring and ONTO the arm

HARTZELL PROPELLER OWNER'S MANUAL 147

- (11) Safety all mounting bolts with 0.032 inch (0.81 mm) minimum diameter stainless steel wire or equivalent aircraft safety cable. (Two bolts per safety.)
- (12)Decompress the external beta system and remove the beta ring puller TE100.

CAUTION

THE BETA RING MUST NOT CONTACT ANY ENGINE COMPONENT OR MOUNTING BOLT SAFETY WIRE. THE BETA MECHANISM FEEDBACK COULD BE DAMAGED IF IT CONTACTED ANY STATIC ENGINE COMPONENT WHILE ROTATING.

- (13)Examine the beta ring to make sure that it is not in contact with any engine components or mounting bolt safety wire.
 - (a) If there is contact between the beta ring and any engine components or mounting bolt safety wire, consult a certified propeller repair station with the appropriate rating.
- (14)Install the beta feedback block assembly into the beta linkage lever, in accordance with the airframe manufacturer's instructions, if applicable.

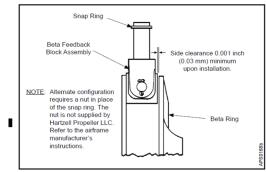
CAUTION:

FIT THE BETA FEEDBACK BLOCK ASSEMBLY IN THE BETA RING WITH A MINIMUM SIDE CLEARANCE OF 0.001 INCH (0.03 mm). REFER TO FIGURE 3-6.

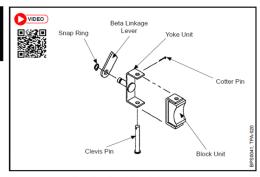
(15)Install the beta feedback block assembly into the beta ring, if applicable. Refer to Figure 3-7.

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HARTZELL PROPELLER OWNER'S MANUAL 147



Beta Feedback Block Assembly and Beta Ring Clearance Figure 3-6



Beta Feedback Block Assembly Figure 3-7

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2. Cessna Maintenance Manual 61-10-00 and 61-20-00

NOTE: The carbon block initially supplied with each propeller has been prefit. If a different carbon block is being installed, it may be necessary to sand it to obtain a total clearance between carbon block and side of groove of 0.001 to 0.002 inch at the tightest point.

(15) Install carbon block (3) onto reversing lever (1).

NOTE: The lower end of the propeller reversing lever is machined with a stepped notch.

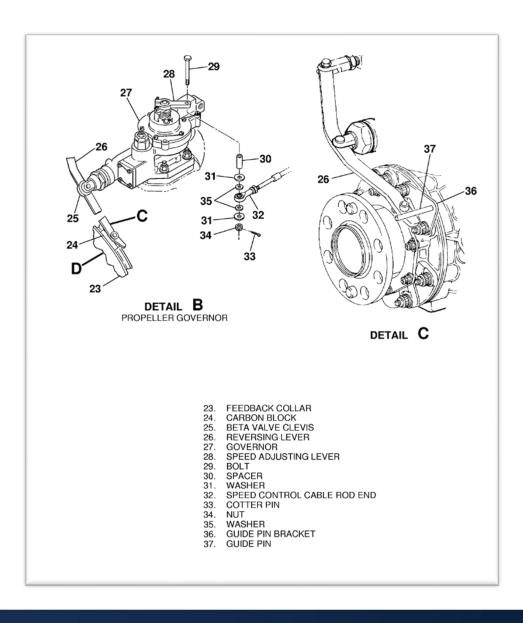
AUTION: Make sure the stepped notch at the end of the propeller reversing lever (1) is under the guide pin (10) in the reversing lever guide pin bracket (15)

(16) Install reversing lever (1) to Beta valve clevis (2) and follow-up ring (4). For the installation procedures for the propeller reversing lever refer to the applicable Pratt & Whitney Engine Maintenance Manual found in the Introduction List of Publications

Position spinner (15) to spinner bulkhead (22) as indexed during removal procedure and secure with screws and fiber washers (5).

- (18) Install right nose cap.
- (19) Check clearance between spinner (15) and nose cap, clearance should be 0.32 inch, +0.10 or -0.10 inch.
- (20) For procedures to install and adjust TKS system propeller components, refer to Chapter 30, TKS Anti-Ice Propeller (McCauley) Maintenance Practices

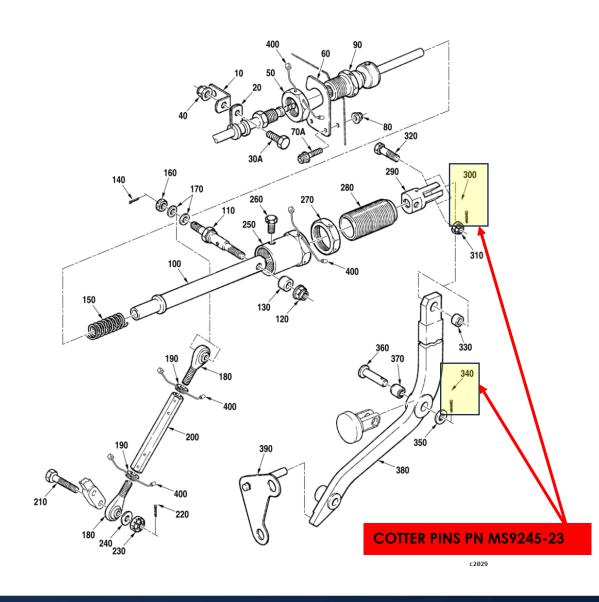
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3. Pratt and Whitney Parts Manual (PT6A-114A) 76-10-00 Figure 2

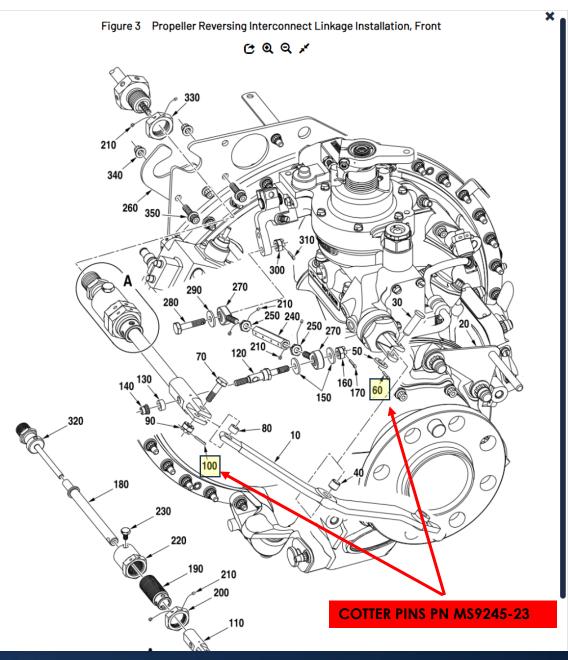
300	MS9245-23	• PIN, COTTER	A,B,C,D	1
310	3011976	NUT, CASTELLATED, HEX INTRCHG WITH P/N MS9358-09		1
_ 310A	MS9358-09	• NUT, CASTELLATED, HEX	A,B,C,D	1
320	3013166	BOLT, MACHINE, HEX	A,B,C,D	1
330	3009087	• SPACER, SLEEVE	A,B,C,D	1
340	MS9245-23	• PIN, COTTER	A,B,C,D	1





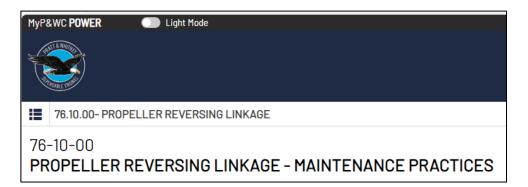
3. Pratt and Whitney Parts Manual (PT6A-140) 76-10-00 Figure 3

	อบ	MS93ZI-10	• WASHER, FLAT		1
	60	MS9245-23	- PIN, COTTER		1
	70	3013166	BOLT, MACH, HEX, DRILLED		1
	80	3009087	• SPACER, SLEEVE		1
	90	MS9358-09	• NUT, CASTLE, HEX		1
1	100	MS9245-23	• PIN, COTTER		1
	110	7017900	OLEVIO DODIEND		1





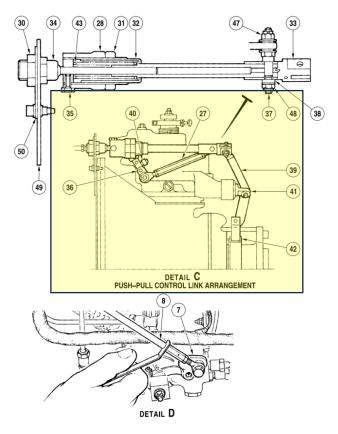
3. Pratt and Whitney Maintenance Manual (PT6A-114A) 76-10-00, Par 5 (B,2,f-h)



(f) Install the sleeve spacer in the hole at the upper end of the propeller reversing lever (39) and fit the lever into the rod end clevis (33). Secure with a bolt and castellated nut. Tighten the nut 12 to 18 lb.in., and lock with a cotterpin.

CAUTION: THE LOWER END OF THE PROPELLER REVERSING LEVER (39) IS MACHINED WITH A STEPPED NOTCH. MAKE SURE THE STEPPED NOTCH AT THE END OF THE PROPELLER REVERSING LEVER (39) IS UNDER THE PIN IN THE REVERSING LEVER GUIDE PIN BRACKET ASSEMBLY (53) (REF. FIG. 201, DETAIL E).

- (g) Install the sleeve bushing in the center hole of the propeller reversing lever and locate the lever in the clevis of the Beta control valve (41). Secure the lever with a straight headed pin, washer and cotterpin.
- (h) With the propeller feathered and the carbon block (42) resting against the rear face of the propeller feedback ring, adjust the low pitch adjuster stop (32) so that, with the linkage pulled fully forward, the Beta valve clevis slot end is flush with the valve capnut. Tighten the locknut (31) 150 to 250 lb.in., and secure the locknut to the stop adjuster (28) with lockwire.





3. Pratt and Whitney Maintenance Manual (PT6A-140) 76-10-01, Par 6, (E, 1-11)



Subtask 76-10-01-400-002

E. Installation of the Front Linkage

(Ref. Fig. 206)

- (1) Install the front linkage assembly on the lifting bracket (16). Attach it with the retaining plate (19), the two bolts (18) and the two nuts (17). Torque the nuts 27 to 30 lbf.in. (3.1-3.3 Nm).
- (2) Install the nut (15) on the front swivel joint (2). Torque the nut 95 to 105 lbf.in. (10.6-12.1 Nm).

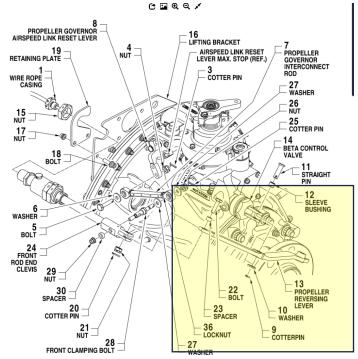
CAUTION: THE LOWER END OF THE PROPELLER REVERSING LEVER (13) HAS A MACHINED STEPPED NOTCH. MAKE SURE THAT THE STEPPED NOTCH OF THE LEVER IS BELOW THE GUIDE PIN IN THE REVERSING LEVER GUIDE PIN BRACKET ASSEMBLY (37).

- (3) Install the sleeve bushing (12) into the middle hole of the propeller reversing lever (13). Put the lever in fork of the beta control valve (14).
- (4) Attach the propeller reversing lever (13) in the beta control valve (14) with the pin (11), the washer (10). Lock the straight pin with the cotterpin (9).
- (5) Install the adjustment fixture (PWC50956) on propeller shaft. Install the fixture ball lockpin to hold the propeller control lever (13).
- (6) Adjust the linkage until a slave bolt goes through freely the rear end of rod assembly (7) and propeller governor airbleed link (8).
- (7) Remove the slave bolt, then turn the two rod ends of the rod assembly one half turn counterclockwise to decrease the overhaul length of the rod. (8) Remove the adjustment fixture (PWC50956) from propeller shaft.
- (9) Apply light grease (PWC04-001) to the rod end of the interconnect rod (7). Attach the interconnect rod to the air bleed link reset lever (8) with the bolt (5), the washer (6), the nut (4).
- (10) Torque the nut (4) 24 to 36 lbf.in. (2.8-4.1Nm). Lock the nut with cotterpin (3).
- (11) Torque the two locknuts (36) 12 to 18 lbf.in. (1.4-2.0 Nm).

WARNING: PUT ON EYE PROTECTION WHEN YOU INSTALL THE SAFETY CABLE. PIECES CAN BREAK OFF AND CAUSE INJURY

(12) Install safety cable (PWC05-344) on the two locknuts (36) and the nut (15) with the crimper (PWC90025).

- (1) Remove all tools, equipment and items not necessary from the work area.
- (2) Install the wire rope casing (Ref. Task 76-10-01-400-802).





108053 DEICE (BRUSH BLOCK) INSTALLATION (IF APPLICABLE)



KIT 108053 AIRFRAME DEICE INSTALLATION

	<u>Hartzell P</u>	ropeller Overhaul Kits	
<u>Overhaul Kit</u> 1	08053	<u>Description</u> AIRFRAME DE-ICE KIT	<u>Rev</u>
Component	Descript	<u>ion</u>	Quantit
B-3837-N832	WASHER	, CORROSION RESISTANT	6
B-3855-31	WASHER	,LOCK,EXTERNAL TOOTH	2
B-6637-51	SCREW,	PAN HEAD, CRES.	2
B-6637-52	SCREW,	PAN HEAD, CRES.	2
B-6655-08	NUT, HE	X, SELF-LOCKING	4
1H1157	SHIM, BI	RUSH BLOCK	1
105273	MOV MO	DDULE - ASSEMBLY	1
108054	BRACKET	, MOUNTING, BRUSH BLOCK	1
3H2090-1	PROPERTY AND PARTY OF THE PROPERTY OF THE PROPERTY OF THE PARTY OF THE	AR BRUSH BLOCK ASSEMBLY	1





USING HARTZELL ICE PROTECTION SYSTEM MANUAL 180, KIT PN 108053 CAN BE FOUND UNDER INSTALLATION INSTRUCTIONS 12AE. NOTE THAT IT SENDS YOU TO STC HOLDER. REFERENCE WIPAIRE DRAWING 1012329 (REV B AND LATER).

HARTZELL ICE PROTECTION SYSTEM MANUAL 180

This section includes the parts list(s) and installation instructions for the following airframe de-ice kit(s): 106095 and 108053

E. Installation Instruction 12AE

 Install the airframe de-ice kit components in accordance with the applicable aircraft TC or STC holder's ICA.

FIG./ITEM PART DESCRIPTION UPA					
NUMBER	NUMBER	DESCRIPTION	OLA	O/H	
	106095 105070 1H1157 3H2090-1 106343 B-3837-N832 B-3855-31 B-6637-51 B-6637-52 B-6655-08	AIRFRAME DE-ICE KIT INSTALLATION INSTRUCTION 12AE BRACKET, MOUNTING SHIM, BRUSH BLOCK MODULAR BRUSH BLOCK ASSEMBLY MOV MODULE - ASSEMBLY WASHER, CORROSION RESISTANT WASHER, LOCK, EXTERNAL TOOTH SCREW, 8-32, PAN HEAD, CRES NUT, HEX, SELF-LOCKING	2 2 2 2 12 4 4 4 8	Y Y	
	108053 108054 1H1157 3H2090-1 105273 B-3837-N832 B-3855-31 B-6637-51 B-6637-52 B-6655-08	AIRFRAME DE-ICE KIT INSTALLATION INSTRUCTION 12AE BRACKET, MOUNTING, BRUSH BLOCK SHIM, BRUSH BLOCK MODULAR BRUSH BLOCK ASSEMBLY MOV MODULE - ASSEMBLY WASHER, CORROSION RESISTANT WASHER, LOCK, EXTERNAL TOOTH SCREW, 8-32, PAN HEAD, CRES SCREW, 8-32, PAN HEAD, CRES NUT, HEX, SELF-LOCKING	1 1 1 6 2 2 2 4	Y Y	
ITEM NOV	ILLUSTRATED				

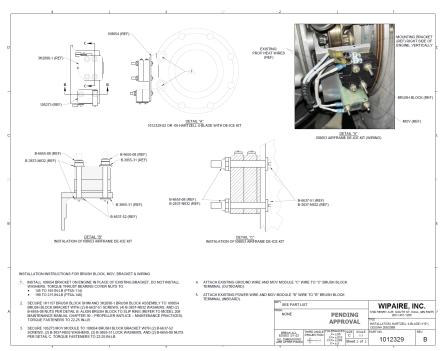
Airframe De-ice Kit(s): 106095

AIRFRAME DE-ICE INSTALLATION and PARTS 30-61-80 Page 12AE-1 Rev. 39 Dec/22



DEICE BRUSH BLOCK BRACKET INSTALLATION.

REFERENCE WIPAIRE DRAWING 1012329 (REV B OR LATER) FOR INSTALLATION AND BUILD UP REFERENCE



108054 BRACKET

AIRFRAME POWER WIRES



MODULAR BRUSH BLOCK ASSEMBLY

MOV MODULE ASSEMBLY

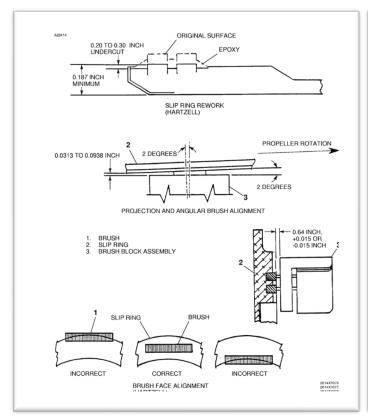


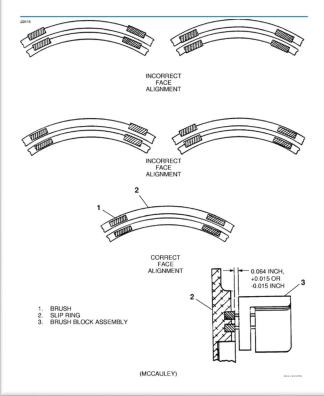
ALIGN BRUSH BLOCK USING HARTZELL ICE PROTECTION MANUAL 180, FIGURE 7-1 AND 208 MM, 30-60-00, PAR 11, FIGURE 202 AS SHOWN BFI OW

11. Brush Block Assembly to Slip Ring Alignment (Hartzell)

CAUTION: Ensure that slip ring alignment has been accomplished before attempting to align brushes on slip ring

- A. Align Brush Block Assembly to Slip Ring Attachment (Refer to Figure 202).
 - NOTE: Keep brushes retracted in brush block until slip ring and propeller assemblies have been installed. In order to get smooth, efficient and quiet transfer of electric power from brushes to slip ring, brush alignment must be checked and adjusted, to meet the following requirements.
 - (1) The clearance between brush block (3) and slip ring (2) must be 0.064 inch, +0.015 or -0.015 inch.
 - (2) The brushes are to be lined up with slip ring so that entire face of each brush (1) is in contact with slip ring (2) throughout the full 360 degrees of slip ring rotation
 - (3) The brushes must contact slip ring at an angle of 2 degree from perpendicular to slip ring surface, measured toward the direction of rotation of slip ring.
 - (4) Brush projection can be adjusted by loosening hardware attaching the brush block and holding the brushes in desired location while retightening hardware. Slotted holes are provided.
 - (5) To center brushes on slip ring, a shim made of a series of laminates is provided and may be peeled for proper alignment. Layers of metal 0.003 inch are used to make up shims which are approximately 0.20 thick overall. Shims may also be fabricated locally.





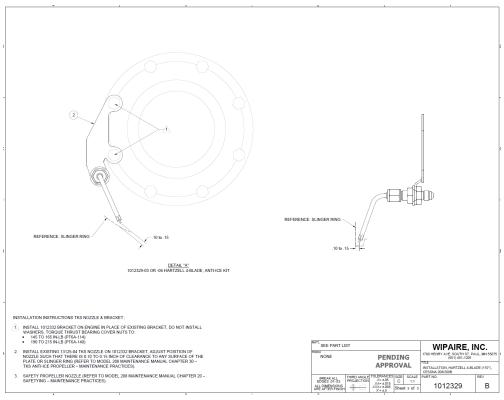


ANTI ICE (TKS) INSTALLATION (IF APPLICABLE)



TKS BRACKET INSTALLATION.

REFERENCE WIPAIRE DRAWING 1012329 (REV B OR LATER) FOR INSTALLATION AND BUILD UP REFERENCE



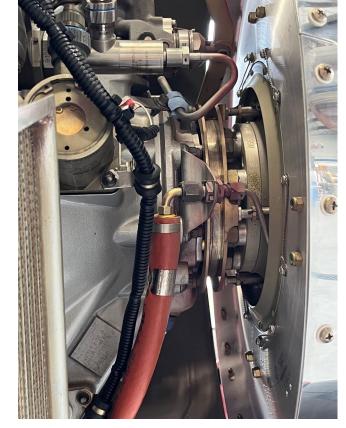
WIPAIRE 1012332 BRACKET



IT IS POSSIBLE THAT THE AIRFRAME HAS THE MCCAULEY 4HFR34C778 PROPELLER INSTALLED PER STC SA09850AC FROM **ALAMO AEROSPACE/MCCAULEY**. REMOVE PROPELLER PER ALAMO AEROSPACE ICA 55181-6.

THE BRACKET PN B-40763 WOULD BE REMOVED AND WIPAIRE BRACKET 1012332 BE INSTALLED. TKS NOZZLE TO BE INSTALLED AS ORIGINAL. STC TKS NOZZLE MAY NOT BE COMPATIBLE WITH HARTZELL INSTALLATION.





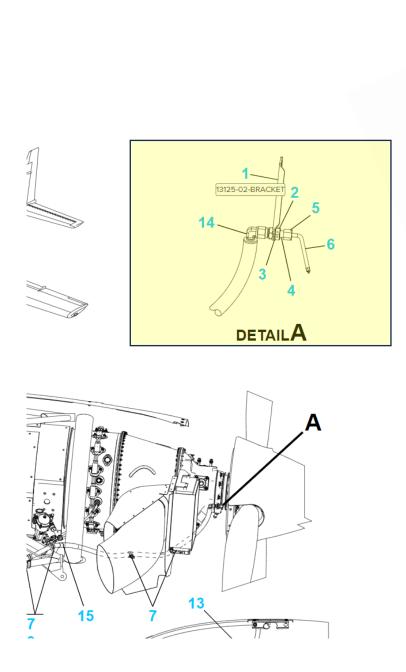


B-40763
BRACKET - TKS NOZZLE



ASSEMBLE TKS NOZZLE ONTO WIPAIRE BRACKET.

REFERENCE WIPAIRE DRAWING 1012329 (REV B OR LATER) FOR INSTALLATION AND BUILD UP REFERENCE







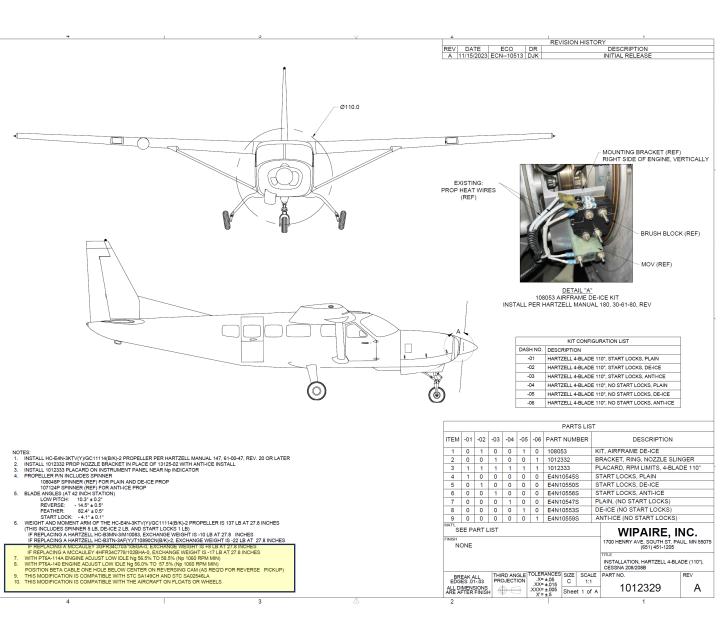
ENGINE RIGGING HIGHLIGHTS



THIS OPERATION WILL REQUIRE AN ADVANCED AMOUNT OF KNOWLEDGE FOR RIGGING THE PT6 TO FACTORY SPECIFICATIONS. THE YUKON PROPELLER INSTALLATION CHANGES A SINGLE RIG POINT, HOWEVER THAT CHANGE MAY AFFECT OTHER RIGGING POINTS. BE PREPARED AND FAMILIARIZE YOURSELF WITH RIGGING A PT6 (-114A AND -140 AS APPLICABLE) PRIOR TO THIS PORTION OF YOUR YUKON PROPELLER INSTALLATION.



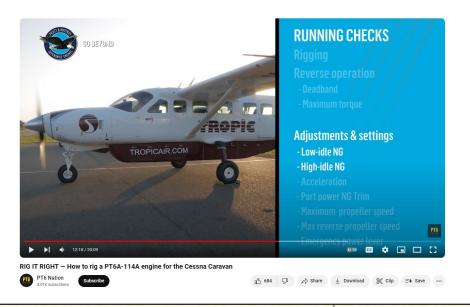
WIPAIRE INSTALLATION DRAWING 1012329



Pay particular attention of notes shown on LH side of drawing for rigging points.



If the assumption is made that the engine is properly rigged prior to previous propeller removal, only one parameter is required to be changed on the PT6A-114A installation. This can be found in the video below, reference the 12:18 point of the video. NOTE: By adjusting the low idle NG per note 7, other parameters of the engine rigging may change. Be wary of completing a single adjustment and expecting all other rigging to hold properly.



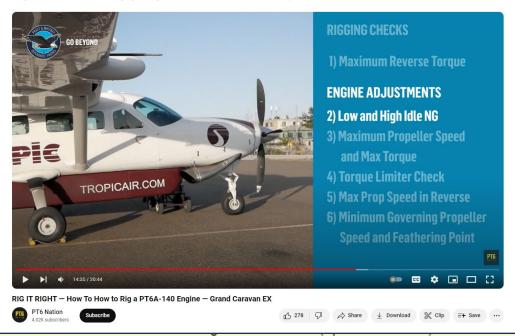
WITH PT6A-114A ENGINE ADJUST LOW IDLE Ng 56.5% TO 58.5% (Np 1060 RPM MIN)

https://www.youtube.com/watch?v=H7c2NiFCNHo



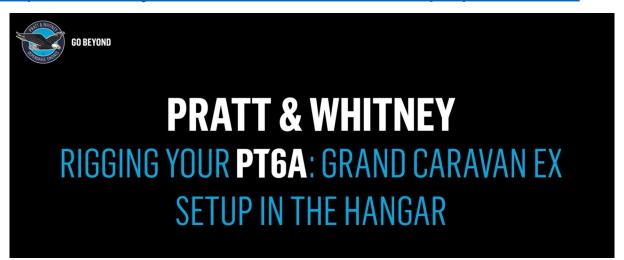


If the assumption is made that the engine is properly rigged prior to previous propeller removal, only one parameter is required to be changed on the PT6A-140 installation. This can be found in the -140 video below, reference the *14:35* point of the video. NOTE: By adjusting the low idle NG per note 8, other parameters of the engine rigging may change. Be wary of completing a single adjustment and expecting all other rigging to hold properly.



WITH PT6A-140 ENGINE ADJUST LOW IDLE Ng 56.0% TO 57.5% (Np 1060 RPM MIN)
 POSITION BETA CABLE ONE HOLE BELOW CENTER ON REVERSING CAM (AS REQ'D FOR REVERSE PICKUP)

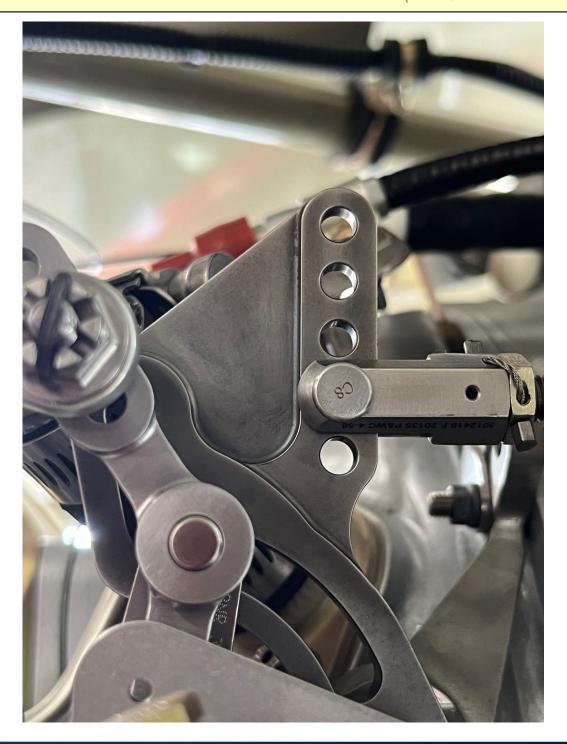
https://www.youtube.com/watch?v=0pApxImLiMw





Reversing cam reference as noted below.

POSITION BETA CABLE ONE HOLE BELOW CENTER ON REVERSING CAM (AS REQ'D FOR REVERSE PICKUP)



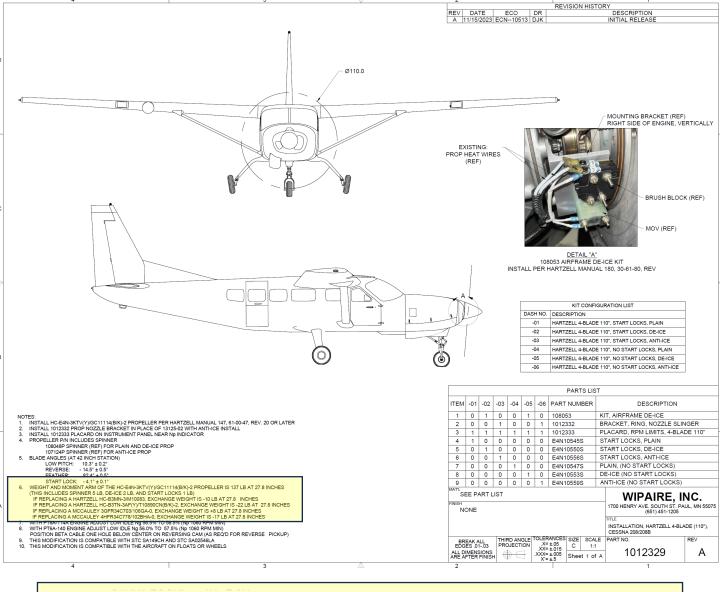


PERFORM FULL ENGINE RUNS WITH RIGGING CHECKS USING VIDEOS SHOWN AS A GUIDE. ENSURE THAT ALL STOCK PARAMATERS ARE BEING HIT INCLUDING THE INCREASED LOW IDEL NG PER WIPAIRE DRAWING 1012329 FOR YOUR INSTALLED ENGINE (-140 OR -114A) AND RIG AND ADJUST ACCORDINGLY. PERFORM INSPECTION AND ENSURE ALL SAFETY WIRE, COTTER PINS, ETC... ARE REINSTALLED UPON RIGGING COMPLETION



WEIGHT AND BALANCE NOTATIONS

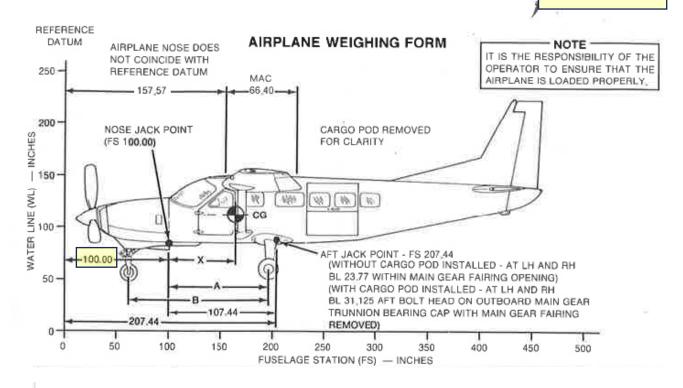




6. WEIGHT AND MOMENT ARM OF THE HC-E4N-3KTV(Y)/GC11114(B/K)-2 PROPELLER IS 137 LB AT 27.8 INCHES (THIS INCLUDES SPINNER 5 LB, DE-ICE 2 LB, AND START LOCKS 1 LB)
IF REPLACING A HARTZELL HC-B3MN-3/M10083, EXCHANGE WEIGHT IS -10 LB AT 27.8 INCHES IF REPLACING A HARTZELL HC-B3TN-3AF(Y)/T10890CN(B/K)-2, EXCHANGE WEIGHT IS -22 LB AT 27.8 INCHES IF REPLACING A MCCAULEY 3GFR34C703/106GA-0, EXCHANGE WEIGHT IS +8 LB AT 27.8 INCHES IF REPLACING A MCCAULEY 4HFR34C778/102BHA-0, EXCHANGE WEIGHT IS -17 LB AT 27.8 INCHES

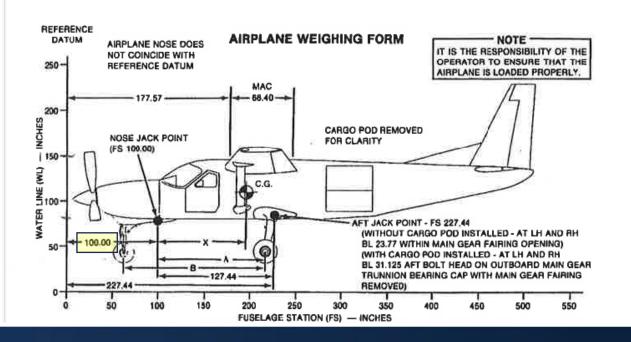
NOTE: WEIGHTS AND ARMS APPLY TO BOTH THE 208B GRAND CARAVAN AND THE SHORTER 208. BOTH OF THE 208 CARAVAN MODELS HAVE A DATUM THAT IS 100" AHEAD OF THE FIREWALL AND RESULTING ARMS ARE NOT AFFECTED BY EITHER MODEL







CESSNA MODEL 208B





END

